

LAUNCHER AND SPACECRAFT CONTROLS

Minotaur V Launch Pad:

M = Move mobile service tower

Automatic illumination at night, extra launch smoke effects, auto self-delete after launch.

Attachment point at base of launcher.

Launcher released from pad when main engine reaches 80% thrust.

Minotaur V Stage 1-3:

Stages 1-3 are controlled by "multistage2.dll" and have the normal controls:

F = Jettison Fairing

J = Jettison stage/payload

P = Ascent Auto-pilot (includes auto fairing/stage separation)

Press P to start the auto-pilot countdown at T-10 seconds

Auto-pilot ends after Stage3 burnout.

Minotaur V Stage 4:

J = Jettison stage

Burn to depletion solid-fuel motor, no throttle. Rotation RCS only.

No auto-pilot.

dV info on HUD.

Minotaur V Stage 5 "Star37FM":

J = Jettison payload

N = Ignite spin-motors

M = Deploy de-spin yoyos

Burn to depletion solid-fuel motor, no throttle. Rotation RCS only.

No auto-pilot.

dV info on HUD.

Use de-spin yoyos to decrease rotation after burn.

LADEE:

455N main engine, four 22N RCS thrusters.

Rotation RCS is unbalanced (rotation will give small dV in +Z direction) and does NOT work well with standard Orbiter autopilots Prograde/Retrograde/OrbitNormal.

Linear RCS in +Z direction only, dV info on HUD.

LADEE also has Reaction Wheels for attitude control

Use [Shift]+[NumPad] controls to use Reaction Wheels:

[Shift]+[NumPad 2/8] = Pitch

[Shift]+[NumPad 1/3] = Yaw

[Shift]+[NumPad 4/6] = Bank

[Shift]+[NumPad 5] = KillRot

Reaction Wheel "KillRot" function status on HUD.

FLIGHT INFO

Launch: Wallops Island 07 September 2013, 03:27:00 UTC

Initial orbit 200km x 271000km, 37.7deg inc(equ), AgP 155deg

Gravity assist from Moon at 1st apogee - boosts perigee to aprox.2000km

Three phasing loops before final TCM burn at perigee to intercept Moon (should be possible to go direct without phasing loops)

LOI: 06 October 2013, 10:57 UTC

Final lunar science orbit 20km x 60km, 157deg inc(equ)

MORE INFO.....

<http://www.nasa.gov/sites/default/files/files/LADEE-Press-Kit-08292013.pdf>

<http://www.orbital.com/NewsInfo/MissionUpdates/MinotaurV/files/LADEE-Mission-Overview.pdf>

FLIGHT NOTES

Press P at T-10 to start ascent autopilot for Stages 1-3.

The autopilot has an increasing throttle curve for Stage 1 to simulate changing motor ISP and gives a better fit to the nominal ascent profile.

Autopilot ends after burnout of Stage 3, you should be sub-orbital with apogee of aprox.260km.

Begin Stage 4 prograde burn at apogee to raise perigee to 206km, apogee 550km, AgP 155deg. Perigee should be on next node.

Begin Star 37FM (Stage 5) prograde burn just before perigee. I used IMFD "DeltaVelocity" and IMFD "Map[Plan View]" to set the time of the burn, targeting the lunar gravity assist at 1st apogee - this should pull the next perigee up to about 2000km alt.

After 1st apogee, IMFD "Course - Target Intercept" together with IMFD "Map[Plan View]" was used to set the final perigee burn to intercept the Moon. Small burns at the next two perigees can optimize the final perigee TLI burn considerably.

IMFD "DeltaVelocity" was used for the final LOI burns.

Have fun :-)

BrianJ

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